

We note that herein (since $F_k = 0$) the resultants N , M and Q are linear functions of the real loads F_i ($i = 1, \dots, n$), whereas N , M and Q are the resultants due to virtual unit forces $F_k = 1$.

Using Eqs. (6.12) and (6.13), we may further differentiate the above relation with respect to F_k , and get

$$\delta_{ik} = \int_L \left\{ \frac{N_i N_k}{EA} + \frac{M_i M_k}{EI} + \frac{Q_i Q_k}{GA} \right\} dr. \quad (6.51)$$

where now N_i , M_i and Q_i are the resultants due to unit forces $F_i = 1$. This important relation makes it possible to calculate the influence coefficients δ_{ik} from product integrals of the given kind. A brief compilation of product integrals, covering the most commonly encountered functions, is given in Table 6.2.

With the help of these influence coefficients, we may reformulate Eq. (6.50) to give

$$f_k = \sum_i \delta_{ki} F_i. \quad (6.52)$$

Furthermore, we also may use these coefficients to solve statically indeterminate systems. For a system of degree k , we introduce the k redundants as external forces X_i on the primary structure, and thus get from Eq. (6.52)

$$0 = \delta_{k0} + \sum_i \delta_{ki} X_i, \quad (6.53)$$

where the δ_{k0} describe the deformations at point k of the primary structure under the given load system.

If all redundants X_i are determined from the above system of equations, we finally can describe the distributions of the resultants as, e.g. for the bending moment M ,

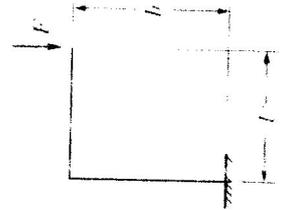
$$M = M_0 + \sum_i M_i X_i, \quad (6.54)$$

where M_0 is the bending moment of the primary structure under the given load system.

Example 6.1:

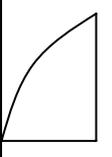
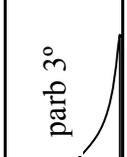
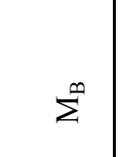
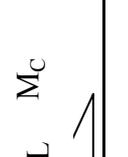
The plane frame shown in the figure has a fixed support and carries a vertical load F at the free end. Both members of the frame have constant flexural rigidity EJ .

Determine the horizontal deflection δ_h , the vertical deflection δ_v , and the angle of rotation γ .



	lk		$\frac{1}{2}lk$		$\frac{1}{2}l(k_1 + k_2)$
	$\frac{1}{2}lk$		$\frac{1}{3}lk$		$\frac{1}{6}l(k_1 + 2k_2)$
	$\frac{1}{2}lk$		$\frac{1}{6}lk$		$\frac{1}{6}l(2k_1 + k_2)$
	$\frac{1}{2}l(k_1 + k_2)$		$\frac{1}{6}l(1 + 2k_2)$		$\frac{1}{6}l(2k_1 + k_2 + 2k_2)$
	$\frac{2}{3}lk$		$\frac{1}{3}lk$		$\frac{1}{3}l(k_1 + k_2)$
	$\frac{2}{3}lk$		$\frac{1}{4}lk$		$\frac{1}{12}l(5k_1 + 3k_2)$
	$\frac{2}{3}lk$		$\frac{5}{12}lk$		$\frac{1}{12}l(3k_1 + 5k_2)$
	$\frac{1}{3}lk$		$\frac{1}{4}lk$		$\frac{1}{12}l(k_1 + 3k_2)$
	$\frac{1}{3}lk$		$\frac{1}{12}lk$		$\frac{1}{12}l(3k_1 + k_2)$
	$\frac{1}{2}lk$		$\frac{1}{6}l(1 + \alpha)k$		$\frac{1}{6}l\{1 + \beta\}k_1 + (1 + \alpha)k_2$
$\int_0^l [M_k(x)]^2 dx$	lk^2		$\frac{1}{3}lk^2$		$\frac{1}{3}l(k_1^2 + k_1 k_2 + k_2^2)$

Table 6.2 Product integrals $\int M_i M_k dr$

EXPRESIONES DE LAS INTEGRALES : $\int_0^L M ds$		EJ = CTE.	L = LONGITUD DE BARRA
M_A parb 3° 	$3/4 M_A \overline{M} L$	$3/10 M_A \overline{M} L$	$3/20 M_A (3\overline{M}_A + 2\overline{M}_B) L$
M_A 	$1/4 M_A \overline{M} L$	$1/20 M_A \overline{M} L$	$1/20 M_A (4\overline{M}_A + \overline{M}_B) L$
parb 3° M_B 	$1/4 M_B \overline{M} L$	$1/5 M_B \overline{M} L$	$1/20 M_B (\overline{M}_A + 4\overline{M}_B) L$
αL 	$1/2 M_C \overline{M} L$	$1/6 (1 + \alpha) M_C \overline{M} L$	$1/6 M_C ((1 + \beta) \overline{M}_A + (1 + \alpha) \overline{M}_B) L$
$\int \overline{M}^2 ds$ 	$\overline{M}^2 L$	$1/3 \overline{M}^2 L$	$1/3 (\overline{M}_A^2 + \overline{M}_B^2 + \overline{M}_A \overline{M}_B) L$